

$\lambda_1, m_1, n_1$ ;  $\lambda_2, m_2, n_2$ ;  $\lambda_3, m_3, n_3$  respectively, such that

$$\left. \begin{aligned} \lambda_1 \tau_{xx} + m_1 \tau_{yy} + n_1 \tau_{zz} &= \lambda_1 \sigma_1 \\ \lambda_1 \tau_{xy} + m_1 \tau_{yx} + n_1 \tau_{zy} &= m_1 \sigma_1 \\ \lambda_1 \tau_{xz} + m_1 \tau_{yz} + n_1 \tau_{zx} &= n_1 \sigma_1 \end{aligned} \right\} \dots \dots (4)$$

$$\left. \begin{aligned} \lambda_2 \tau_{xx} + m_2 \tau_{yy} + n_2 \tau_{zz} &= \lambda_2 \sigma_2 \\ \lambda_2 \tau_{xy} + m_2 \tau_{yx} + n_2 \tau_{zy} &= m_2 \sigma_2 \\ \lambda_2 \tau_{xz} + m_2 \tau_{yz} + n_2 \tau_{zx} &= n_2 \sigma_2 \end{aligned} \right\} \dots \dots (5)$$

and another set of equations is obtained by replacing  $\lambda_2, m_2, n_2$  and  $\sigma_2$  by  $\lambda_3, m_3, n_3$  and  $\sigma_3$ .

Now we shall prove that the three principal stress values  $\sigma_1, \sigma_2, \sigma_3$  are real and that the corresponding principal stress directions are mutually orthogonal.

Multiply the three equations of (4) by  $\lambda_2, m_2, n_2$  respectively and add,

$$\begin{aligned} \lambda_1 \lambda_2 \tau_{xx} + m_1 m_2 \tau_{yy} + n_1 n_2 \tau_{zz} + (\lambda_2 m_1 + \lambda_1 m_2) \tau_{xy} + (m_2 n_1 + m_1 n_2) \tau_{yz} \\ + (n_2 \lambda_1 + n_1 \lambda_2) \tau_{zx} = \sigma_1 (\lambda_1 \lambda_2 + m_1 m_2 + n_1 n_2) \end{aligned}$$

[considering the symmetry of  $\tau_{xy} = \tau_{yx}$  etc.]

similarly, multiply the three equations of (5) by  $\lambda_1, m_1, n_1$  respectively and add,

$$\begin{aligned} \lambda_1 \lambda_2 \tau_{xx} + m_1 m_2 \tau_{yy} + n_1 n_2 \tau_{zz} + (\lambda_2 m_1 + \lambda_1 m_2) \tau_{xy} + (m_2 n_1 + m_1 n_2) \tau_{yz} \\ + (n_2 \lambda_1 + n_1 \lambda_2) \tau_{zx} = \sigma_2 (\lambda_1 \lambda_2 + m_1 m_2 + n_1 n_2) \end{aligned}$$

subtracting these two equations, we get

$$0 = (\sigma_1 - \sigma_2) (\lambda_1 \lambda_2 + m_1 m_2 + n_1 n_2) \dots \dots (6)$$

similarly, we can obtain

$$0 = (\sigma_2 - \sigma_3) (\lambda_2 \lambda_3 + m_2 m_3 + n_2 n_3) \dots \dots (7)$$

$$\text{and } 0 = (\sigma_1 - \sigma_3) (\lambda_3 \lambda_1 + m_3 m_1 + n_3 n_1) \dots \dots (8)$$

Let us now consider eq<sup>n</sup> (6) which is a cubic eq<sup>n</sup> in  $\sigma$  with roots  $\sigma_1, \sigma_2$  and  $\sigma_3$ . Let us now assume that this equation has a complex root. Since this is a cubic equation with real coefficients so another root must also be complex which will be the complex conjugate of the former. So the set of roots may be written as  $\sigma_1 = \alpha + i\beta, \sigma_2 = \alpha - i\beta, \sigma_3$  where  $\alpha, \beta$  and  $\sigma_3$  are real numbers. So equation (6) can be written as

$$\left. \begin{aligned} \lambda_2 \bar{\tau}_{xx} + m_2 \bar{\tau}_{yy} + n_2 \bar{\tau}_{zz} &= \lambda_2 \bar{\sigma}_1 \\ \lambda_2 \bar{\tau}_{xy} + m_2 \bar{\tau}_{yx} + n_2 \bar{\tau}_{zy} &= m_2 \bar{\sigma}_1 \\ \lambda_2 \bar{\tau}_{xz} + m_2 \bar{\tau}_{yz} + n_2 \bar{\tau}_{zx} &= n_2 \bar{\sigma}_1 \end{aligned} \right\} \dots \dots (9)$$

Since  $\sigma_2 = \bar{\sigma}_1$  and  $\tau_{2x}, \tau_{2y}$  etc. being real,  $\tau_{2x} = \bar{\tau}_{1x}, \tau_{2y} = \bar{\tau}_{1y}$  etc. the coefficients of  $\lambda^2, m^2, n^2$  in  $e^{\lambda^2}(9)$  are complex conjugate of the coefficients of  $\lambda_1, m_1, n_1$  in  $e^{\lambda^2}(9)$ . So the values of  $\lambda_2, m_2, n_2$  determined from equation (9) are the complex conjugate of the values of  $\lambda_1, m_1, n_1$  determined from (9). So

if  $\lambda_1 = a_1 + ib_1, m_1 = a_2 + ib_2, n_1 = a_3 + ib_3$  then

$\lambda_2 = a_1 - ib_1, m_2 = a_2 - ib_2, n_2 = a_3 - ib_3$

$\therefore \lambda_1 \lambda_2 + m_1 m_2 + n_1 n_2 = (a_1 + ib_1)(a_1 - ib_1) + (a_2 + ib_2)(a_2 - ib_2) + (a_3 + ib_3)(a_3 - ib_3)$   
 $= a_1^2 + b_1^2 + a_2^2 + b_2^2 + a_3^2 + b_3^2 \neq 0$ .

$\therefore$  it follows that from  $e^{\lambda^2}(6)$  thus  $\sigma_1 - \sigma_2 = 0$

$\therefore \lambda + i\beta - \lambda + i\beta = 0$

or  $2i\beta = 0$

$\Rightarrow \beta = 0$ .

This contradicts the original assumption that the roots are complex. So the assumption of the existence of complex root of  $e^{\lambda^2}(3)$  is not true, i.e. the roots  $\sigma_1, \sigma_2, \sigma_3$  are all real.

If  $\sigma_1 \neq \sigma_2 \neq \sigma_3$ , then from  $e^{\lambda^2}(6), (7), (8)$  we find that the principal stress directions are mutually orthogonal. If  $\sigma_1 = \sigma_2 \neq \sigma_3$  then  $\lambda_3, m_3, n_3$  are fixed but we can not determine an infinite number of values of the d-cs  $(\lambda_1, m_1, n_1)$  and  $(\lambda_2, m_2, n_2)$  orthogonal to  $\lambda_3, m_3, n_3$ .

If  $\sigma_1 = \sigma_2 = \sigma_3$  then any set of orthogonal axes may be taken as principal axes.

## Stress Invariants © 992

Let  $Ox, Oy, Oz$  be a system of orthogonal axes w.r.t. which the stress tensor is

$$\begin{bmatrix} \tau_{xx} & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \tau_{yy} & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \tau_{zz} \end{bmatrix}$$

Let  $\vec{r}$  be the direction of principal stress at O. Then

$\vec{r}$  is in the direction of  $\vec{r}$ . If  $\lambda, m, n$  be the d-cs of this line and if  $\sigma$  be the magnitude of principal stress then

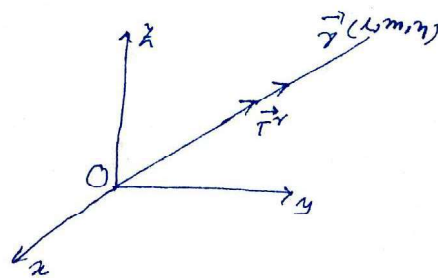
$\tau_{yx} = \lambda\sigma, \tau_{xy} = m\sigma, \tau_{xz} = n\sigma$

i.e.

$\lambda \tau_{xx} + m \tau_{yx} + n \tau_{zx} = \lambda\sigma$  or  $\lambda(\tau_{xx} - \sigma) + m \tau_{yx} + n \tau_{zx} = 0$

$\lambda \tau_{xy} + m \tau_{yy} + n \tau_{zy} = m\sigma$   $\lambda \tau_{xy} + m(\tau_{yy} - \sigma) + n \tau_{zy} = 0$

$\lambda \tau_{xz} + m \tau_{yz} + n \tau_{zz} = n\sigma$   $\lambda \tau_{xz} + m \tau_{yz} + n(\tau_{zz} - \sigma) = 0$



eliminating  $x, y, z$ ; we have

$$\begin{vmatrix} \tau_{xx} - \sigma & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \tau_{yy} - \sigma & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \tau_{zz} - \sigma \end{vmatrix} = 0$$

this is a cubic eq<sup>n</sup> in  $\sigma$ .

this can be written as,

$$\sigma^3 - (\oplus)\sigma^2 + H\sigma - \Delta = 0 \quad \dots (A)$$

where  $\oplus = \tau_{xx} + \tau_{yy} + \tau_{zz} \dots (1)$

$$H = \begin{vmatrix} \tau_{xx} & \tau_{yx} \\ \tau_{xy} & \tau_{yy} \end{vmatrix} + \begin{vmatrix} \tau_{xx} & \tau_{xz} \\ \tau_{zx} & \tau_{zz} \end{vmatrix} + \begin{vmatrix} \tau_{yy} & \tau_{yz} \\ \tau_{zy} & \tau_{zz} \end{vmatrix} \quad \dots (2)$$

All the three roots of the eq<sup>n</sup> (A) are real and they are the values of three principal stresses  $\sigma_1, \sigma_2, \sigma_3$ .  
 since the principal stresses characterised the physical state of stress at a point they are independent of any co-ordinate of reference. therefore

(a)  $\tau_{xx} + \tau_{yy} + \tau_{zz}$  being equal to  $\sigma_1 + \sigma_2 + \sigma_3$  is invariant under a co-ordinate transformation (1st Invariant).

(b)  $\begin{vmatrix} \tau_{xx} & \tau_{yx} \\ \tau_{xy} & \tau_{yy} \end{vmatrix} + \begin{vmatrix} \tau_{xx} & \tau_{xz} \\ \tau_{zx} & \tau_{zz} \end{vmatrix} + \begin{vmatrix} \tau_{yy} & \tau_{yz} \\ \tau_{zy} & \tau_{zz} \end{vmatrix}$  being equal to

$\sigma_1\sigma_2 + \sigma_2\sigma_3 + \sigma_3\sigma_1$  is also invariant under co-ordinate transformation (2nd Invariant).

(c)  $\begin{vmatrix} \tau_{xx} & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \tau_{yy} & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \tau_{zz} \end{vmatrix}$  being equal to  $\sigma_1\sigma_2\sigma_3$  is also invariant under co-ordinate transformation (3rd Invariant).

**SUM** If the state of stress at any point of a body be given by  $\tau_{xx} = \sigma_x = y^2 + \gamma(x^2 - y^2)$

$$\tau_{yy} = \sigma_y = x^2 + \gamma(y^2 - x^2) \quad \dots (1)$$

$$\tau_{zz} = \sigma_z = x^2 + y^2$$

$\tau_{yz} = \tau_{zy} = 0$  and  $\tau_{xy} = f(x, y)$  determine the expression for  $\tau_{xy}$  in order that the stress distribution is in equilibrium in the absence of body force.

Sol<sup>n</sup>: Eqs of equilibrium are (in absence of body force)

$$\frac{\partial}{\partial x} \tau_{xx} + \frac{\partial}{\partial y} \tau_{yx} + \frac{\partial}{\partial z} \tau_{zx} = 0 \quad \dots (1)$$

$$\frac{\partial}{\partial x} \tau_{xy} + \frac{\partial}{\partial y} \tau_{yy} + \frac{\partial}{\partial z} \tau_{zy} = 0 \quad \dots (2)$$

$$\frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z} = 0 \quad \dots (3)$$

From the given condition we know from (1) and (2),

$$2\gamma x + \frac{\partial f(x,y)}{\partial y} = 0 \quad \dots (4)$$

$$2\gamma y + \frac{\partial f(x,y)}{\partial x} = 0 \quad \dots (5)$$

'Eq' (3) is identically satisfied.

From (4)  $\frac{\partial f(x,y)}{\partial y} = -2\gamma x$

Int. w.r.t,  $f(x,y) = -2\gamma xy + f_1(x) \quad \dots (6)$

From (5),  $\frac{\partial f(x,y)}{\partial x} = -2\gamma y$

Int. w.r.t,  $f(x,y) = -2\gamma xy + f_2(y) \quad \dots (7)$

From (6) and (7)  $f_1(x) = f_2(y) = c$  where  $c$  is any constant

$$\tau_{xy} = f(x,y) = -2\gamma xy + c$$

### Stress quadric of Cauchy, 293

Let  $ox, oy, oz$  be a set of rectangular axes through  $O$ . Let us consider a quadric surface  $\tau_{xx}x^2 + \tau_{yy}y^2 + \tau_{zz}z^2 + 2\tau_{yz}yz + 2\tau_{zx}zx + 2\tau_{xy}xy = \pm k^2$  where  $k$  is constant and  $\tau_{xx}, \tau_{yy}$  etc. are the components of stress tensor at  $O$  referred to  $ox, oy$  and  $oz$  as axes. This is called the stress quadric of Cauchy.

Let us make a transformation of axes according to the following scheme

	$x$	$y$	$z$
$x'$	$l_1$	$m_1$	$n_1$
$y'$	$l_2$	$m_2$	$n_2$
$z'$	$l_3$	$m_3$	$n_3$

The eq<sup>n</sup> of quadric (1) referred to these new set of axes  $ox', oy', oz'$  becomes

$$\tau_{xx}(l_1x' + l_2y' + l_3z')^2 + \tau_{yy}(m_1x' + m_2y' + m_3z')^2 + \tau_{zz}(n_1x' + n_2y' + n_3z')^2 + 2\tau_{yz}(m_1x' + m_2y' + m_3z')(n_1x' + n_2y' + n_3z') + 2\tau_{zx}(n_1x' + n_2y' + n_3z')(l_1x' + l_2y' + l_3z') + 2\tau_{xy}(l_1x' + l_2y' + l_3z')(m_1x' + m_2y' + m_3z') = \pm k^2 \quad \dots (2)$$

The co-efficient of  $x'^2$  in L.H.S. of the above eq<sup>n</sup> is

$$\tau_{xx}l_1^2 + \tau_{yy}m_1^2 + \tau_{zz}n_1^2 + 2\tau_{yz}m_1n_1 + 2\tau_{zx}n_1l_1 + 2\tau_{xy}l_1m_1 = \tau_{x'x'}$$

Similarly, the co-efficient of  $y'^2$  and  $z'^2$  are  $\tau_{y'y'}$  and  $\tau_{z'z'}$  respectively. [By eq<sup>n</sup> of stress transformation law]

The co-efficient of  $2x'y'$  of (2) is

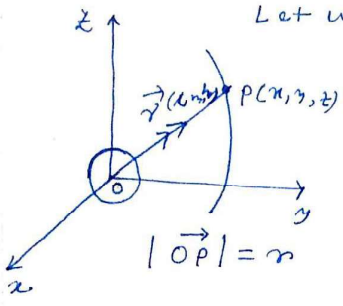
$$\tau_{xx}l_1l_2 + \tau_{yy}m_1m_2 + \tau_{zz}n_1n_2 + (m_2n_1 + m_1n_2)\tau_{yz} + (n_2l_1 + m_1l_2)\tau_{zx} + (l_2m_1 + l_1m_2)\tau_{xy} = \tau_{x'y'}$$

Similarly the co-efficient of  $2y'z'$  and  $2z'x'$  are respectively  $\tau_{y'z'}$  and  $\tau_{z'x'}$ . So the eq<sup>n</sup> of the quadric surface given by (1) when referred to  $ox', oy', oz'$  as axes  $\rightarrow$

takes the form

$$\tau_{xx}x^2 + \tau_{yy}y^2 + \tau_{zz}z^2 + 2\tau_{yz}yz + 2\tau_{zx}zx + 2\tau_{xy}xy = \pm k^2 \quad (3)$$

Now eq<sup>n</sup> of the stress quadric referred to its principal axes as the co-ordinate axes  $OX, OY, OZ$  will obviously be of the form  $\tau_{xx}x^2 + \tau_{yy}y^2 + \tau_{zz}z^2 = \pm k^2$  showing that the principal axes of the stress quadric are also the principal axes of stress at  $O$ .



Let us now study the other properties of the quadric surface given by eq<sup>n</sup> (1). We draw any unit area through  $O$ . Let  $\vec{n}$  be the unit normal to this elementary area where d-cts of  $\vec{n}$  are  $l, m, n$ . Let us draw the radius vector  $\vec{OP}$  to the quadric in this direction so that  $P(x, y, z)$  is a point on the quadric and  $|\vec{OP}| = r, \therefore \frac{x}{r} = l, \frac{y}{r} = m, \frac{z}{r} = n$ .

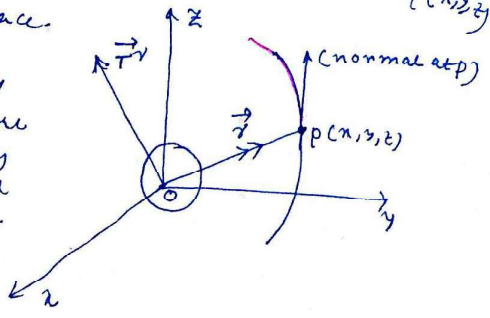
The stress exerted by the material on the side towards which normal  $\vec{n}$  is drawn on the opposite side across the unit area has for its component  $\tau_{xx}, \tau_{yy}, \tau_{zz}$ . The components of the stress in the direction of  $\vec{n}$  i.e. the normal component of stress is given by,

$$\begin{aligned}
 l\tau_{xx} + m\tau_{yy} + n\tau_{zz} &= l(l\tau_{xx} + m\tau_{yx} + n\tau_{zx}) + m(m\tau_{xy} + m\tau_{yy} + n\tau_{zy}) \\
 &\quad + n(n\tau_{xz} + m\tau_{yz} + n\tau_{zz}) \\
 &= l^2\tau_{xx} + m^2\tau_{yy} + n^2\tau_{zz} + 2lm\tau_{xy} + 2mn\tau_{yz} + 2nl\tau_{zx} \\
 &= \frac{1}{r^2} [x^2\tau_{xx} + y^2\tau_{yy} + z^2\tau_{zz} + 2xy\tau_{xy} + 2yz\tau_{yz} + 2zx\tau_{zx}] \\
 &= \pm \frac{k^2}{r^2} [LHS]
 \end{aligned}$$

So, if the quadric surface with centre at  $O$  given by eq<sup>n</sup> (1) can be drawn then the normal stress on any unit area through  $O$  can easily be derived by drawing the normal to the unit area. If the normal cuts the quadric surface at a distance  $r$  from  $O$  then  $\pm \frac{k^2}{r^2}$  is the normal stress.

Another interesting property of the quadric surface given by eq<sup>n</sup> (1) is that the normal to this surface at the end of the radius vector  $\vec{OP}$  is in the direction parallel to the stress vector  $\vec{T}$  at  $O$  where  $\vec{OP}$  is in the direction of unit vector  $\vec{n}$  and  $P(x, y, z)$  is a point on the quadric surface.

To prove this property we firstly note that the stress exerted by the material on the side of  $\vec{n}$  across unit area at  $O$  of which normal to  $\vec{n}$  to the material on the other side is  $\vec{T}$ . Its components are  $\tau_{xx}, \tau_{yy}, \tau_{zz}$ .



$\therefore$  d-cts of  $\vec{T}$  are  $\tau_{xx}, \tau_{yy}, \tau_{zz}$ . Now eq<sup>n</sup> of the quadric surface

(1) can be written as

$$f(x,y,z) = \tau_{xx}x^2 + \tau_{yy}y^2 + \tau_{zz}z^2 + 2\tau_{yz}yz + 2\tau_{zx}zx + 2\tau_{xy}xy - FK^2 = 0$$

Dir of the normal to the surface at P

$$\frac{\partial f}{\partial x} = \tau_{xx} \cdot 2x + 2\tau_{zx}z + 2\tau_{xy}y = 2(\tau_{xx}x + \tau_{zx}z + \tau_{xy}y)$$

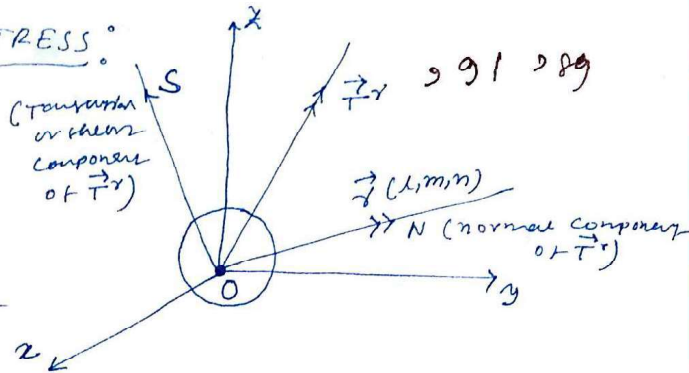
$$\frac{\partial f}{\partial x} = 2r(\tau_{xx}x + m\tau_{zx}z + n\tau_{xy}y) = 2r\tau_{xx}$$

$$\frac{\partial f}{\partial y} = 2r\tau_{xy} \quad , \quad \text{similarly,} \quad \frac{\partial f}{\partial z} = 2r\tau_{yz}$$

This shows that normal at P is parallel to the stress vector  $\vec{T}^x$ . So, if the stress quadric surface (1) be drawn at any point some property of the stress at the centre of the quadric can at once be derived because of these facts the eq<sup>n</sup> (1) is called the eq<sup>n</sup> of stress quadric.

### MAXIMUM SHEARING STRESS:

In order to determine the maximum shearing stress at any point O of the medium, we take the co-ordinate axes at O of  $ox, oy, oz$  which are in the direction of principal stresses at O and let  $\sigma_1, \sigma_2, \sigma_3$  be the corresponding stress values such that



$$\tau_{xx} = \sigma_1, \quad \tau_{yx} = 0, \quad \tau_{zx} = 0$$

$$\tau_{xy} = 0, \quad \tau_{yy} = \sigma_2, \quad \tau_{zy} = 0$$

$$\tau_{xz} = 0, \quad \tau_{yz} = 0, \quad \tau_{zz} = \sigma_3$$

We consider any arbitrary unit area through O with normal  $\vec{n}$  determined by the d-cs  $(l, m, n)$ . Then the stress exerted by the material on the side towards which normal  $\vec{n}$  is drawn to the material on the ~~side towards~~ opposite side across the unit area has for its components  $\tau_{rx}, \tau_{ry}, \tau_{rz}$  where

$$\left. \begin{aligned} \tau_{rx} &= l\tau_{xx} + m\tau_{zx} + n\tau_{yx} \\ \tau_{ry} &= l\tau_{xy} + m\tau_{yy} + n\tau_{zy} \\ \tau_{rz} &= l\tau_{xz} + m\tau_{yz} + n\tau_{zz} \end{aligned} \right\} \dots \dots \dots (1)$$

The resultant stress can be written as

$$(\vec{T}^r)^2 = \tau_{rx}^2 + \tau_{ry}^2 + \tau_{rz}^2 = l^2\sigma_1^2 + m^2\sigma_2^2 + n^2\sigma_3^2 \dots \dots (2)$$

The normal stress N can be written as

$$N = l\tau_{xx} + m\tau_{yy} + n\tau_{zz} = l^2\sigma_1 + m^2\sigma_2 + n^2\sigma_3 \dots \dots (3)$$

If S be the magnitude of the shearing stress, then

$$N^2 + S^2 = (\vec{T}^r)^2$$

$$\therefore S^2 = (\vec{T}^r)^2 - N^2 = (l^2\sigma_1^2 + m^2\sigma_2^2 + n^2\sigma_3^2) - (l^2\sigma_1 + m^2\sigma_2 + n^2\sigma_3)^2 \dots \dots (4)$$

with rotation of the unit area ~~at the point O~~ the shearing stress  $S$  varies, being a function of two variables, say,  $l$  and  $m$ . Since we know,  $n^2 = 1 - l^2 - m^2$ . Using these relations, (4) becomes

$$S^2 = (\sigma_1^2 - \sigma_3^2) l^2 + (\sigma_2^2 - \sigma_3^2) m^2 + \sigma_3^2 - [(\sigma_1 - \sigma_3) l^2 + (\sigma_2 - \sigma_3) m^2 + \sigma_3]^2 \quad (5)$$

To obtain maximum value of shearing stress we equate to zero the partial derivatives of  $S$  with respect to  $l$  and  $m$ . At the values of  $l, m$  for which  $S$  is maximum,  $S^2$  will also be maximum. Therefore we equate  $\frac{\partial S^2}{\partial l} = 0, \frac{\partial S^2}{\partial m} = 0$  from which we obtain,

$$(\sigma_1^2 - \sigma_3^2) l - 2 [(\sigma_1 - \sigma_3) l^2 + (\sigma_2 - \sigma_3) m^2 + \sigma_3] (\sigma_1 - \sigma_3) l = 0 \quad (6)$$

$$(\sigma_2^2 - \sigma_3^2) m - 2 [(\sigma_1 - \sigma_3) l^2 + (\sigma_2 - \sigma_3) m^2 + \sigma_3] (\sigma_2 - \sigma_3) m = 0 \quad (7)$$

consider the most general case when  $\sigma_1, \sigma_2$  and  $\sigma_3$  are all different. then dividing eq<sup>n</sup> (6) by  $(\sigma_1 - \sigma_3)$  and eq<sup>n</sup> (7) by  $(\sigma_2 - \sigma_3)$  we obtain

$$\left\{ (\sigma_1 - \sigma_3) - 2 [(\sigma_1 - \sigma_3) l^2 + (\sigma_2 - \sigma_3) m^2] \right\} l = 0 \quad (8)$$

$$\left\{ (\sigma_2 - \sigma_3) - 2 [(\sigma_1 - \sigma_3) l^2 + (\sigma_2 - \sigma_3) m^2] \right\} m = 0 \quad (9)$$

we have two equations of the third degree in  $l$  and  $m$ . Accordingly we shall obtain three solutions of  $l, m$  and  $n$ . The first and the simplest solution is  $l = 0, m = 0, n = 1$  corresponding to this value of  $l, m$  and  $n$ . We find from equation (5) that  $S = 0$  this verifies the fact that plane element normal to the principal direction of stress is stress free. Thus the minimum value of  $|S|$  is associated with the principal directions. Since we are seeking for maximum shearing stress, so discarding this value of  $l, m, n$  we have other three values (i)  $l \neq 0, m = 0$ , (ii)  $l = 0, m \neq 0$  (iii)  $l \neq 0, m \neq 0$ . The third case is impossible since cancelling  $l$  and  $m$  out of equation (8) and (9) respectively and subtracting the resulting equation we immediately obtained  $\sigma_1 = \sigma_2$  which contradicts our original assumption that  $\sigma_1 \neq \sigma_2 \neq \sigma_3$ .

Next consider the 1st case,  $l \neq 0, m = 0$ ; then from eq<sup>n</sup> (8) we obtained  $(\sigma_1 - \sigma_3) (1 - 2l^2) = 0$  which gives

$$l = \pm \frac{1}{\sqrt{2}}, m = 0, n = \pm \frac{1}{\sqrt{2}} \quad (10)$$

when we consider the 2nd case,  $l = 0, m \neq 0$ , then from eq<sup>n</sup> (9) we obtained  $(\sigma_2 - \sigma_3) (1 - 2m^2) = 0$  which gives

$$l = 0, m = \pm \frac{1}{\sqrt{2}}, n = \pm \frac{1}{\sqrt{2}} \quad (11)$$

If at the end set we had eliminated, say,  $m$  instead of  $n$  from eq<sup>n</sup> (4) and repeated the analysis we would have obtained additionally one more solution,  $l = \pm \frac{1}{\sqrt{2}}, m = \pm \frac{1}{\sqrt{2}}, n = 0$  --- (12)

The extreme values of shear stress can be obtained by substituting these values of  $l, m, n$  in (5) by the help of (10) and (5), we obtain

$$(S_1^2)_{\max} = \frac{\sigma_1^2 + \sigma_3^2}{2} - \frac{(\sigma_1 + \sigma_3)^2}{4} = \frac{2(\sigma_1^2 + \sigma_3^2) - (\sigma_1 + \sigma_3)^2}{4} = \frac{(\sigma_1 - \sigma_3)^2}{4}$$

i.e.  $(S_1)_{\text{extremum}} = \pm \frac{\sigma_1 - \sigma_3}{2}$

Similarly, by the help of (11) and (5)

$$(S_2)_{\text{extremum}} = \pm \frac{\sigma_2 - \sigma_3}{2} \quad \text{and by the help of (12) and (5),}$$

$$(S_3)_{\text{extremum}} = \pm \frac{\sigma_1 - \sigma_2}{2}$$

If  $\sigma_1 > \sigma_2 > \sigma_3$ , then the maximum value of  $|S|$  is

$$|S|_{\max} = \frac{1}{2} (\sigma_1 - \sigma_3)$$

In this case we find from eq<sup>n</sup> (10) that the maximum shearing stress acts on the surface elements containing the  $y$ -principal axis and bisecting the angle between  $x$  and  $z$  axes.

They we conclude that the maximum shearing stress is equal to the one half the difference between the greatest and least normal stresses and acts on the plane that bisects the angle between the directions of the largest and smallest principal stresses.

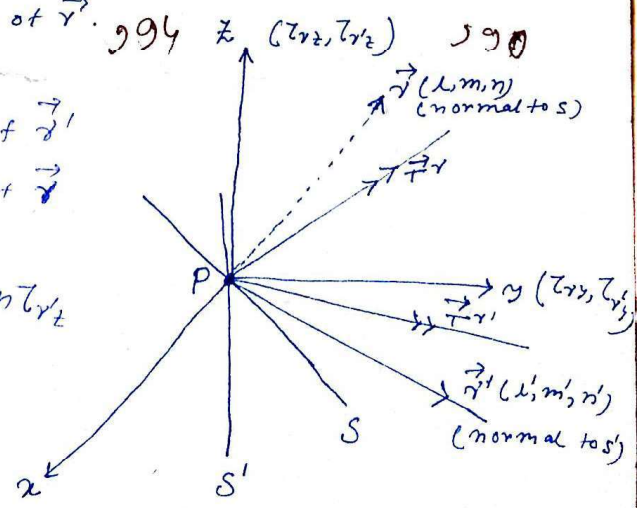
**THEOREM**

Let surface elements  $S$  and  $S'$  with unit normals  $\vec{n}$  and  $\vec{n}'$  pass through the point  $P$ . Show that the component of stress vector  $\vec{T}_x$  (acting on  $S$ ) in the direction of  $\vec{n}'$  is equal to the component of stress vector  $\vec{T}_{x'}$  (acting on  $S'$ ) in the direction of  $\vec{n}$ .

Proof: We have to prove  
 Component of  $\vec{T}_x$  in the direction of  $\vec{n}'$   
 = Component of  $\vec{T}_{x'}$  in the direction of  $\vec{n}$   
 i.e. to prove

$$l'\tau_{xx} + m'\tau_{xy} + n'\tau_{xz} = l\tau_{yx} + m\tau_{yy} + n\tau_{yz}$$

$$\begin{aligned} \text{Now } l\tau_{yx} + m\tau_{yy} + n\tau_{yz} \\ = l(l'\tau_{xx} + m'\tau_{yx} + n'\tau_{zx}) \\ + m(l'\tau_{xy} + m'\tau_{yy} + n'\tau_{zy}) \\ + n(l'\tau_{xz} + m'\tau_{zy} + n'\tau_{zz}) \end{aligned}$$



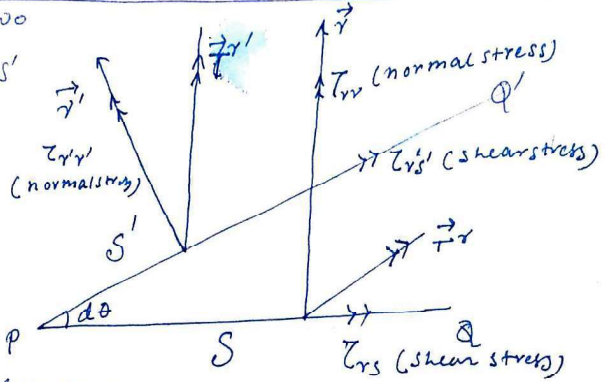


$$\begin{aligned}
 l \tau'_{yx} + m \tau'_{yz} + n \tau'_{zz} &= l' (l \tau_{xx} + m \tau_{yy} + n \tau_{zz}) \\
 &+ m' (l \tau_{xy} + m \tau_{yy} + n \tau_{yz}) \\
 &+ n' (l \tau_{xz} + m \tau_{yz} + n \tau_{zz})
 \end{aligned}$$

$$\Rightarrow l \tau'_{yx} + m \tau'_{yz} + n \tau'_{zz} = l' \tau_{xx} + m' \tau_{yy} + n' \tau_{zz} \text{ [proved]}$$

Hence show that the normal stress was a stationary value (max or min) when the shear stress is zero. 294

Let  $PQ$  and  $PQ'$  are two surface elements  $S$  and  $S'$  whose normal are in the direction of  $\vec{r}$  and  $\vec{r}'$  respectively. Angle between  $PQ$  and  $PQ'$  is a small angle  $d\theta$ .  $\tau_{xy}$  and  $\tau_{yx}$  are



The normal and shear components of stress on  $S$  and  $\tau_{yx}$  and  $\tau'_{yx}$  are the normal and shearing components of stress on  $S'$ . Obviously,  $\tau'_{yx}$  and  $\tau'_{yx}$  are different from  $\tau_{xy}$  and  $\tau_{yx}$  and depend on  $\theta$ . So, let us have the relation as

$$\tau'_{yx} = \tau_{xy} + \frac{\partial \tau_{xy}}{\partial \theta} d\theta \quad \text{and} \quad \tau'_{yx} = \tau_{yx} + \frac{\partial \tau_{yx}}{\partial \theta} d\theta$$

We know that components of  $\vec{F}'$  in the direction  $\vec{r} =$  component of  $\vec{F}$  in the direction of  $\vec{r}$ .

$$\therefore \tau'_{yx} \cos d\theta + \tau'_{yx} \sin d\theta = \tau_{xy} \cos d\theta - \tau_{yx} \sin d\theta$$

$$\begin{aligned}
 \therefore (\tau_{xy} + \frac{\partial \tau_{xy}}{\partial \theta} d\theta) \cos d\theta + (\tau_{yx} + \frac{\partial \tau_{yx}}{\partial \theta} d\theta) \sin d\theta \\
 = \tau_{xy} \cos d\theta - \tau_{yx} \sin d\theta
 \end{aligned}$$

$$\therefore \frac{\partial \tau_{xy}}{\partial \theta} \cos d\theta + \frac{\partial \tau_{yx}}{\partial \theta} \sin d\theta = 2 \tau_{yx} \frac{\sin d\theta}{d\theta}$$

Taking limit as  $d\theta \rightarrow 0$ , we get

$\frac{\partial \tau_{xy}}{\partial \theta} = -2 \tau_{yx}$  i.e. rate of change of normal component of stress with  $\theta$  equals to  $-2 \times$  shearing component of stress.

If  $\tau_{yx} = 0$ , then  $\frac{\partial \tau_{xy}}{\partial \theta} = 0$

i.e.  $\tau_{xy}$  is stationary.

Ex: 1 w.r.t. the frame of reference  $oxyz$ , let the state of stress be

$$[\tau_{ij}] = \begin{bmatrix} 1 & 2 & 1 \\ 2 & 1 & 1 \\ 1 & 1 & 1 \end{bmatrix} \cdot \text{Determine the principal stresses and their associated directions. Also check the invariants of } I_1, I_2, I_3.$$

Solution

For this state of stress

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$$I_1 = \tau_{11} + \tau_{22} + \tau_{33} = 1 + 1 + 1 = 3$$

$$I_2 = \begin{vmatrix} \tau_{11} & \tau_{21} \\ \tau_{12} & \tau_{22} \end{vmatrix} + \begin{vmatrix} \tau_{22} & \tau_{32} \\ \tau_{23} & \tau_{33} \end{vmatrix} + \begin{vmatrix} \tau_{11} & \tau_{31} \\ \tau_{13} & \tau_{33} \end{vmatrix}$$

$$= \begin{vmatrix} 1 & 2 \\ 2 & 1 \end{vmatrix} + \begin{vmatrix} 1 & 1 \\ 1 & 1 \end{vmatrix} + \begin{vmatrix} 1 & 1 \\ 1 & 1 \end{vmatrix} = -3$$

$$I_3 = \begin{vmatrix} 1 & 2 & 1 \\ 2 & 1 & 1 \\ 1 & 1 & 1 \end{vmatrix} = 0 - 2(1) + 1(1) = -1$$

Let  $\sigma$  be the principal stress value, then the eq<sup>n</sup> for determining

$$\sigma \text{ is } \begin{vmatrix} 1-\sigma & 2 & 1 \\ 2 & 1-\sigma & 1 \\ 1 & 1 & 1-\sigma \end{vmatrix} = 0$$

$$a, \quad \sigma^3 - I_1 \sigma^2 + I_2 \sigma - I_3 = 0$$

$$b, \quad \sigma^3 - 3\sigma^2 - 3\sigma + 1 = 0$$

$$c, \quad (\sigma^3 + 1) - 3\sigma(\sigma + 1) = 0$$

$$d, \quad (\sigma + 1)(\sigma^2 - 4\sigma + 1) = 0$$

$$\Rightarrow \sigma_1 = -1, \quad \sigma_2 = 2 + \sqrt{3}, \quad \sigma_3 = 2 - \sqrt{3}$$

With a set of axes chosen along the principal axes the stress matrix will have the form

$$[\tau'_{ij}] = \begin{bmatrix} -1 & 0 & 0 \\ 0 & 2 + \sqrt{3} & 0 \\ 0 & 0 & 2 - \sqrt{3} \end{bmatrix} \quad \text{Hence } I_1' = -1 + 2 + \sqrt{3} + 2 - \sqrt{3} = 3$$

$$I_2' = \begin{vmatrix} -1 & 0 \\ 0 & 2 + \sqrt{3} \end{vmatrix} + \begin{vmatrix} 2 + \sqrt{3} & 0 \\ 0 & 2 - \sqrt{3} \end{vmatrix} + \begin{vmatrix} -1 & 0 \\ 0 & 2 - \sqrt{3} \end{vmatrix}$$

$$= -2 - \sqrt{3} + 4 - 3 - 2 + \sqrt{3} = -3$$

$$I_3' = \begin{bmatrix} -1 & 0 & 0 \\ 0 & 2 + \sqrt{3} & 0 \\ 0 & 0 & 2 - \sqrt{3} \end{bmatrix} = (-1)(2 + \sqrt{3})(2 - \sqrt{3}) = -1$$

This checks the invariants of  $I_1, I_2$  and  $I_3$ .

Direction of principal stresses  
 If  $l_1, m_1, n_1$  be the d-cs of principal axis corresponding to the principal stress  $\sigma_1 = -1$ , then

$$\begin{cases} l_1 + 2m_1 + n_1 = -1 \cdot l_1 \\ 2l_1 + m_1 + n_1 = -1 \cdot m_1 \\ l_1 + m_1 + n_1 = -1 \cdot n_1 \end{cases} \quad \left| \begin{array}{l} l_1 \tau_{xx} + m_1 \tau_{yx} + n_1 \tau_{zx} = l_1 \sigma_1 \\ l_1 \tau_{yx} + m_1 \tau_{yy} + n_1 \tau_{zy} = m_1 \sigma_1 \\ l_1 \tau_{xz} + m_1 \tau_{yz} + n_1 \tau_{zz} = n_1 \sigma_1 \end{array} \right.$$

$$\Rightarrow 2l_1 + 2m_1 + n_1 = 0$$

$$+ \quad 2l_1 + m_1 + n_1 = 0$$

$$( \quad \quad \quad l_1 + m_1 + 2n_1 = 0$$

( solving the last two equations, we have

$$\frac{l_1}{3} = \frac{m_1}{-3} = \frac{n_1}{0} \quad \Rightarrow \quad \frac{l_1}{1} = \frac{m_1}{-1} = \frac{n_1}{0} = \frac{1}{\sqrt{2}}$$

$$\sigma_1, \quad l_1 = \frac{1}{\sqrt{2}}, \quad m_1 = -\frac{1}{\sqrt{2}}, \quad n_1 = 0$$

{ Again if  $l_2, m_2, n_2$  be the d-cs of the principal axis corresponding to the principal stress  $\sigma_2 = 2 + \sqrt{3}$  then

$$l_2 + 2m_2 + n_2 = l_2 (2 + \sqrt{3})$$

$$+ \quad 2l_2 + m_2 + n_2 = m_2 (2 + \sqrt{3})$$

$$+ \quad \quad \quad l_2 + m_2 + n_2 = n_2 (2 + \sqrt{3})$$

$$\Rightarrow \quad l_2 (-1 - \sqrt{3}) + 2m_2 + n_2 = 0$$

$$2l_2 + (-1 - \sqrt{3})m_2 + n_2 = 0$$

$$+ \quad \quad \quad l_2 + m_2 + (-1 - \sqrt{3})n_2 = 0$$

( cross-multiplication of the last two eqs, we have

$$\frac{l_2}{3 + \sqrt{3}} = \frac{m_2}{3 + \sqrt{3}} = \frac{n_2}{2\sqrt{3}} \quad \Rightarrow \quad \frac{l_2}{\sqrt{3} + 1} = \frac{m_2}{\sqrt{3} + 1} = \frac{n_2}{2} = \frac{1}{\sqrt{12 + 4\sqrt{3}}}$$

$$\therefore l_2 = \frac{1}{2} \left(1 + \frac{1}{\sqrt{3}}\right)^{\frac{1}{2}}$$

$$m_2 = \frac{1}{2} \left(1 + \frac{1}{\sqrt{3}}\right)^{\frac{1}{2}}$$

$$n_2 = \frac{1}{(3 + \sqrt{3})^{\frac{1}{2}}}$$

Now, if  $\vec{N}_1 = (l_1, m_1, n_1)$ ,  $\vec{N}_2 = (l_2, m_2, n_2)$  and

$$\vec{N}_3 = (l_3, m_3, n_3)$$

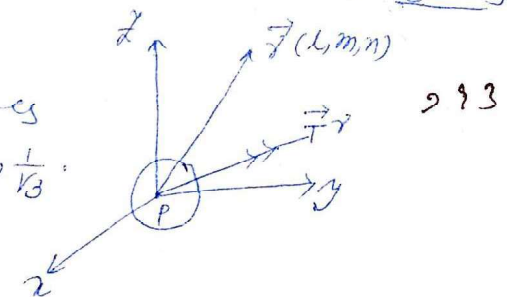
$$\text{then } \vec{N}_3 = \vec{N}_1 \times \vec{N}_2$$

$$\vec{N}_3 = \begin{vmatrix} \vec{i} & \vec{j} & \vec{k} \\ \frac{1}{\sqrt{2}} & -\frac{1}{\sqrt{2}} & 0 \\ \frac{1}{2} \left(1 + \frac{1}{\sqrt{3}}\right)^{1/2} & \frac{1}{2} \left(1 + \frac{1}{\sqrt{3}}\right)^{1/2} & \frac{1}{(3 + \sqrt{3})^{1/2}} \end{vmatrix}$$

$$= \vec{i} \left[ -\frac{1}{\sqrt{2} (3 + \sqrt{3})^{1/2}} \right] + \vec{j} \left[ -\frac{1}{\sqrt{2} (3 + \sqrt{3})^{1/2}} \right] + \vec{k} \left[ \frac{1}{\sqrt{2}} \left(1 + \frac{1}{\sqrt{3}}\right)^{1/2} \right]$$

Ex: 2 At a point P in a body  $\sigma_x = 10000 \text{ kgf/cm}^2$ ,  $\sigma_y = -5000 \text{ kgf/cm}^2$  and  $\tau_{xy} = \tau_{yz} = \tau_{zx} = 10000 \text{ kgf/cm}^2$ . Determine the normal and shearing stresses on a plane which is equally inclined to all the three axes.

Sol: If  $\vec{n}$  be the unit normal to the unit area at P, the cosines of the normal  $\vec{n}$  are  $\frac{1}{\sqrt{3}}, \frac{1}{\sqrt{3}}, \frac{1}{\sqrt{3}}$ .



Now  $\tau_{rx} = l\tau_{xx} + m\tau_{yx} + n\tau_{zx}$

$$= \frac{1}{\sqrt{3}} (\sigma_x + \tau_{yx} + \tau_{zx}) = \frac{1}{\sqrt{3}} (10000 + 10000 + 10000)$$

$$= 10000\sqrt{3}$$

$$\tau_{ry} = l\tau_{xy} + m\tau_{yy} + n\tau_{zy} = \frac{1}{\sqrt{3}} (\tau_{xy} + \sigma_y + \tau_{zy})$$

$$= \frac{1}{\sqrt{3}} (10000 - 5000 + 10000)$$

$$= 5000\sqrt{3}$$

$$\tau_{rz} = l\tau_{xz} + m\tau_{yz} + n\tau_{zz} = \frac{1}{\sqrt{3}} (\tau_{xz} + \tau_{yz} + \sigma_z)$$

$$= \frac{1}{\sqrt{3}} (10000 + 10000 - 5000)$$

$$= 5000\sqrt{3}$$

$$l^2 + m^2 + n^2 = 1$$

$$l = m = n = \frac{1}{\sqrt{3}}$$

If  $\sigma_r$  be the normal component of stress vector on the unit area at P, of which the unit normal is  $\vec{n}$ , then

$$\sigma_r = l\tau_{rx} + m\tau_{ry} + n\tau_{rz} = \frac{1}{\sqrt{3}} (10000\sqrt{3} + 5000\sqrt{3} + 5000\sqrt{3})$$

$$= 20000$$

also  $|\vec{T}_r|^2 = \tau_{rx}^2 + \tau_{ry}^2 + \tau_{rz}^2 = 3 \times 10^8 + 3 \times 25 \times 10^6 + 3 \times 25 \times 10^6$

$$= 450 \times 10^6$$

If  $\theta$  be the angle between the direction of  $\vec{T}_r$  and  $\vec{n}$ , then  $\sigma_r = T^r \cos \theta$  and shearing or tangential stress  $\tau_y = T^r \sin \theta$

$$\sigma_y^2 + \tau_y^2 = T^2$$

$$\Rightarrow \tau_y^2 = T^2 - \sigma_y^2$$

$$\tau_y = \sqrt{T^2 - \sigma_y^2}$$

$$= \sqrt{450 \times 10^6 - 400 \times 10^6}$$

$$= 5000 \sqrt{2}$$

Ex: 3 (Home Task)

The stress tensor at a point P is given by w.r.t. the axes  $ox_1x_2$

by the values  $\tau_{ij} = \begin{bmatrix} 3 & 1 & 1 \\ 1 & 0 & 2 \\ 1 & 2 & 0 \end{bmatrix}$  determine the principal stress values and their direction w.r.t. the axes  $ox_1x_2$ .

$\Rightarrow$  Let  $\sigma$  be the principal stress values, then the eq<sup>n</sup> for determining  $\sigma$  is

$$\begin{vmatrix} 3-\sigma & 1 & 1 \\ 1 & -\sigma & 2 \\ 1 & 2 & -\sigma \end{vmatrix} = 0$$

$$\Rightarrow (3-\sigma)(\sigma^2-4) - (-\sigma-2) + (2+\sigma) = 0$$

$$\Rightarrow (\sigma+2) \{ (3-\sigma)(\sigma-2) + 1 + 1 \} = 0$$

$$\Rightarrow (\sigma+2) (3\sigma - 6 - \sigma^2 + 2\sigma + 2) = 0$$

$$\Rightarrow (\sigma+2) (5\sigma - 4 - \sigma^2) = 0$$

$$\Rightarrow (\sigma+2) (\sigma-4)(1-\sigma) = 0$$

$$\Rightarrow \sigma_1 = -2, \sigma_2 = 4, \sigma_3 = 1$$

If  $l_1, m_1, n_1$  be the d-cs of principal axis corresponding to the principal stress  $\sigma_1 = -2$ , then

$$5l_1 + m_1 + n_1 = 0$$

$$l_1 + 2m_1 + 2n_1 = 0$$

$$l_1 + 2m_1 + 2n_1 = 0$$

$\Rightarrow$  Cross-multiplication of the above two equations

$$\frac{l_1}{0} = \frac{m_1}{-9} = \frac{n_1}{9} \Rightarrow \frac{l_1}{0} = \frac{m_1}{-1} = \frac{n_1}{1} = \frac{1}{\sqrt{2}}$$

$$\Rightarrow l_1 = 0, m_1 = -\frac{1}{\sqrt{2}}, n_1 = \frac{1}{\sqrt{2}}$$

$$\frac{l_1}{0} = \frac{m_1}{1} = \frac{n_1}{-1} = \frac{1}{\sqrt{2}}$$

$$\Rightarrow l_1 = 0, m_1 = \frac{1}{\sqrt{2}}, n_1 = -\frac{1}{\sqrt{2}}$$

Let  $l_2, m_2, n_2$  are the d.c.s of the principal axes corresponding to the principal stress  $\sigma_2 = 4$ , then

$$-l_2 + m_2 + n_2 = 0$$

$$l_2 - 4m_2 + 2n_2 = 0$$

$$l_2 + 2m_2 - 4n_2 = 0$$

By cross-multiplication of the above two equations, we have

$$\frac{l_2}{6} = \frac{m_2}{3} = \frac{n_2}{3}$$

$$\therefore \frac{l_2}{2} = \frac{m_2}{1} = \frac{n_2}{1} = \frac{1}{\sqrt{6}}$$

$$\therefore l_2 = \frac{2}{\sqrt{6}}, m_2 = \frac{1}{\sqrt{6}}, n_2 = \frac{1}{\sqrt{6}}$$

Now if  $\vec{n}_1 = (l_1, m_1, n_1)$ ,  $\vec{n}_2 = (l_2, m_2, n_2)$  and  $\vec{n}_3 = (l_3, m_3, n_3)$

$$\text{Then } \vec{n}_3 = \vec{n}_1 \times \vec{n}_2 = \begin{vmatrix} \vec{i} & \vec{j} & \vec{k} \\ 0 & -\frac{1}{\sqrt{2}} & \frac{1}{\sqrt{2}} \\ \frac{2}{\sqrt{6}} & \frac{1}{\sqrt{6}} & \frac{1}{\sqrt{6}} \end{vmatrix}$$

$$\frac{2}{\sqrt{6}} l_3 + m_3 + n_3 = 0$$

$$l_3 - m_3 + 2n_3 = 0$$

$$\frac{l_3}{3} = \frac{m_3}{-3} = \frac{n_3}{-3} = \left[ -\frac{1}{\sqrt{12}} - \frac{1}{\sqrt{12}} \right] \vec{i} + \left[ \frac{\sqrt{2}}{\sqrt{6}} \right] \vec{j} + \left[ \frac{\sqrt{2}}{\sqrt{6}} \right] \vec{k}$$

$$\frac{l_3}{1} = \frac{m_3}{-1} = \frac{n_3}{-1} = \frac{1}{\sqrt{3}} = -\frac{1}{\sqrt{3}} \vec{i} + \frac{1}{\sqrt{3}} \vec{j} + \frac{1}{\sqrt{3}} \vec{k}$$

$$l_3 = \frac{1}{\sqrt{3}}, m_3 = -\frac{1}{\sqrt{3}}, n_3 = -\frac{1}{\sqrt{3}}$$

(+) Add After 7.2.96

Geometrical interpretation of infinitesimal strain components

(a) Geometrical interpretation of longitudinal strains:

Let  $(x, y, z)$  be the co-ordinates of P and  $(x+dx, y+dy, z+dz)$  be the co-ordinates of Q before deformation.

Let  $PQ = ds$ , so let  $l, m, n$  be the d.c.s of PQ, then

$$l = \frac{dx}{ds}, m = \frac{dy}{ds}, n = \frac{dz}{ds}$$

where  $ds^2 = dx^2 + dy^2 + dz^2$

When deformed slightly P occupies the position P' with co-ordinates

$(x+u_x, y+u_y, z+u_z)$  and Q occupies the position Q' with co-ordinates

